

ch 2 solutions

①

$$\underline{18)} \quad \frac{nR}{V} = \frac{P_{\text{gauge}} + P_{\text{atm}}}{T} = \frac{2,5 \cdot 10^5 \text{ N/m}^2 + 1,013 \cdot 10^5 \text{ N/m}^2}{308,15 \text{ K}} \\ = 1140,03 \frac{\text{N}}{\text{m}^2 \cdot \text{K}}$$

$$P' = \frac{nR}{V} T' = 1140,03 \frac{\text{N}}{\text{m}^2 \cdot \text{K}} \cdot 233,15 = 2,658 \cdot 10^5 \text{ N/m}^2$$

$$P'_{\text{gauge}} = P' - P_{\text{atm}} = \boxed{1,645 \text{ N/m}^2}$$

$$\underline{31)} \quad \text{a)} \quad \frac{nR}{V} = \frac{P}{T} = \frac{1,40 \cdot 10^7 \text{ N/m}^2}{298,15 \text{ K}} = 4,696 \cdot 10^4 \text{ N/m}^2 \cdot \text{K}$$

$$P' = \frac{nR}{V} T' = 4,696 \cdot 10^4 \text{ N/m}^2 \cdot \text{K} \cdot 194,65 \text{ K} = \boxed{9,140 \cdot 10^6 \frac{\text{N}}{\text{m}^2}}$$

$$\text{b)} \quad P'' = 0,9 P' = \boxed{8,226 \cdot 10^5 \text{ N/m}^2}$$

$$\text{c)} \quad T''' = \frac{P''}{nR/V} = \frac{1,013 \cdot 10^5 \text{ N/m}^2}{4,696 \cdot 10^4 \text{ N/m}^2 \cdot \text{K}} = \boxed{2,157 \text{ K}}$$

d) No. Liquid Helium at 4,15K is achievable only in high-tech labs, and this 2,157K is even colder!

$$\underline{39)} \quad v_{\text{rms}} = \sqrt{\frac{3kT}{m}} = \sqrt{\frac{3 \cdot 1,38 \cdot 10^{-23} \cdot 500}{6,646 \cdot 10^{-27}}} = \boxed{176,5 \text{ m/s}}$$

$$\underline{48)} \quad E = \frac{3}{2} kT \Rightarrow T = \frac{E}{\frac{3}{2} k} = \boxed{3,09 \cdot 10^9 \text{ K}}$$

$$\underline{64)} \quad \text{a)} \quad n_{\text{total}} = \frac{PV}{RT} = 91,39 \text{ moles}$$

$$n_{\text{He}} = 0,79 n_{\text{total}} = \boxed{72,20 \text{ mol}}$$

$$n_{\text{O}_2} = 0,21 n_{\text{total}} = \boxed{19,19 \text{ mol}}$$

b) $\Delta U_{He} = n_{He} C_{v,He} \Delta T = 72.20 \cdot 1.50 R \cdot 4 = 3601.8 J$
 $\Delta U_{O_2} = n_{O_2} C_{v,O_2} \Delta T = 19.19 \cdot 2.53 R \cdot 4 = 1614.7 J$
 $\Delta U_{total} = \Delta U_{He} + \Delta U_{O_2} = \boxed{5217 J}$

65) $\Delta U_{N_2O} + \Delta U_{air} = 0$

$\Rightarrow n_{N_2O} C_{v,N_2O} (T_f - T_{i,N_2O}) = n_{air} C_{v,air} (T_f - T_{i,air})$

$\Rightarrow T_f - T_{i,N_2O} = \underbrace{-\frac{n_{air} C_{v,air}}{n_{N_2O} C_{v,N_2O}}}_{\equiv \alpha} (T_f - T_{i,air})$

$\Rightarrow T_f = -\alpha T_f + \alpha T_{i,air} + T_{i,N_2O}$

$\Rightarrow T_f = \frac{\alpha T_{i,air} + T_{i,N_2O}}{1 + \alpha}$ where $\alpha = \frac{4 \cdot 2.50 \cdot 8.3145}{1 \cdot 30.43} = 2.732$
 $= \boxed{271.5 K}$

66) see example 2.10 in text

$M = 34.1 \text{ g/mol} \Rightarrow m = 5.663 \cdot 10^{-26} \text{ kg}$

$v_1 = \sqrt{\frac{3RT}{M}} = 468.45 \text{ m/s}$

$v_2 = 2v_1 = 936.90 \text{ m/s}$

$\frac{f(v_1)}{f(v_2)} = \left(\frac{v_1}{v_2}\right)^2 e^{-\frac{m}{2kT}(v_1^2 - v_2^2)}$
 $= \boxed{22.56}$

67) Equation 2.15 in text $f(v) dv = \frac{4}{\sqrt{\pi}} \left(\frac{m}{2kT}\right)^{3/2} v^2 e^{-\frac{mv^2}{2kT}} dv$

where $m = 4.652 \cdot 10^{-26} \text{ kg}$ from $28.0134 \text{ g/mol} \div 6.022 \cdot 10^{23} \text{ molecules/mol}$
 and $\Delta v = 1 \text{ m/s}$ and $T = 300 \text{ K}$ and $v = 290 \text{ m/s}$

$\Rightarrow \boxed{0.001576}$

(3)

$$76) a) P = \frac{N}{V} kT = (10^9) (k_B) (2.7K) = \boxed{3.73 \cdot 10^{-17} \text{ N/m}^2}$$

$$b) \# \text{density} = \frac{N}{V} \Rightarrow V = \frac{N}{\# \text{density}} = \frac{6.022 \cdot 10^{23}}{10^6} = \boxed{6.022 \cdot 10^{17} \text{ m}^3}$$

$$c) L = \sqrt[3]{V} = \boxed{844 \text{ km}}$$

$$78) \frac{N}{V} = \frac{P}{kT} = 1.984 \cdot 10^{25} \text{ m}^{-3}$$

$$\rho = (0.78 m_{N_2} + 0.21 m_{O_2} + 0.01 m_{Ar}) \frac{N}{V} = \boxed{0.954 \text{ kg/m}^3}$$

$$\text{where } m_{N_2} = 28.0134 \text{ g/mol} \rightarrow 4.652 \cdot 10^{-26} \text{ kg}$$

$$m_{O_2} = 31.998 \text{ g/mol} \rightarrow 5.314 \cdot 10^{-26} \text{ kg}$$

$$m_{Ar} = 39.948 \text{ g/mol} \rightarrow 6.634 \cdot 10^{-26} \text{ kg}$$